

STEREO INTEGRATION AND TEST FLOW

JOE STAIGER

joseph.staiger@jhuapl.edu

SPACE DEPARTMENT

INTEGRATION AND TEST CONCEPTS

- *TEST PHILOSOPHY WILL BE BASED ON APL DOCUMENT SDO 2387-1, MIL- STD-1540B, AND GEVS-SE Rev A.*
- *WILL TEST AND OPERATE TWO SPACECRAFT AS ONE SPACECRAFT WITH A SIDE 1 AND 2.*
- *EACH SPACECRAFT WILL HAVE ITS OWN GROUND SUPPORT EQUIPMENT (GSE). POWER RACK INTERFACE, RF RACK INTERFACE, AND ANY OTHER GSE SIMULATORS (SOLAR ARRAY, BATTERY CHARGER,ETC)*

INTEGRATION AND TEST CONCEPTS

- *STEREO (2) SPACECRAFT WILL BE INTEGRATED AND TESTED IN APL CLEANROOM FACILITIES AND GSFC FACILITIES.*
- *SUBSYSTEMS AND INSTRUMENTS WILL BE DELIVERED TO THE (2) SPACECRAFT, FLIGHT QUALIFIED WITH THEIR RESPECTIVE GSE'S.*
- *THE MISSION OPERATIONS TEAM WILL BE COMBINED WITH THE I & T TEAM TO PROVIDE SPACECRAFT FAMILIARITY AND OPERATING EXPERIENCE.*

TEST OBJECTIVES

- *TEST THE (2) STEREO SPACECRAFT UNDER AMBIENT CONDITIONS AND SIMULATED ENVIRONMENTS EXPECTED TO OCCUR DURING LAUNCH PREPARATIONS, LAUNCH, AND IN ORBIT.*
- *ESTABLISH CORRECT OPERATIONS OF ALL SUBSYSTEMS AND INSTRUMENTS, WHEN INTERCONNECTED AS A SPACECRAFT, CONFIRMING ALL ELECTRICAL INTERFACES TO BE IN COMPLIANCE WITH SPECIFICATIONS AND/OR INTERFACE CONTROL DOCUMENTS.*

TEST OBJECTIVES

- *VALIDATE SPACECRAFT SYSTEM PERFORMANCE THROUGH TESTING.*
- *VALIDATE ACCEPTABLE QUALITY OF WORKMANSHIP BY DEMONSTRATING PERFORMANCE UNDER MISSION-LEVEL ENVIRONMENTAL STRESS.*
- *DEVELOP A DATA BASE AND EXPERTISE TO UNDERSTAND POSSIBLE MALFUNCTIONS SO REMEDIAL ACTION CAN TAKE PLACE IN A TIMELY MANNER SO AS NOT TO AFFECT THE PROGRAM SCHEDULE.*

*ESTIMATED TIMES TO INTEGRATE AND TEST ALL OF THE (2)
STEREO SPACECRAFT SUBSYSTEMS AND INSTRUMENTS.
USING COMPUTERIZED TEST PROCEDURES*

STEREO 1 & 2 POWER ELECTRONICS

TIME

*POWER ELECTRONICS (PS RELAYS, PEAK POWER
TRACKER, & POWER SHUNT/FUSE BOX)*

4 WKS.

BATTERIES (simulators used during I&T)

SOLAR ARRAYS (solar array simulators used during I & T)

STEREO 1 & 2 C&DH ELECTRONICS AND RF SUBSYSTEM

TELEMETRY INTERFACE

TIME

RECORDERS

6 WKS.

C&DH PROCESSOR

ULTRA STABLE OSC.

X-BAND UPLINK

X-BAND DOWNLINK

RF POWER AMPLIFIER

RF SWITCHES & DIPLEXERS

HIGH-GAIN,MED-GAIN,AND LOW-GAIN ANTENNAS

(WITH HIGH-GAIN ANT. GIMBAL)

STEREO 1 & 2 ATTITUDE GUIDANCE AND CONTROL

<i>ATTITUDE INTERFACE ELECTRONICS</i>	<i>TIME</i>
<i>FLIGHT COMPUTER</i>	<i>4 WKS.</i>
<i>(3) REACTION WHEELS & ELECTRONICS</i>	
<i>INERTIAL MEASUREMENT UNIT (GYROS)</i>	
<i>STAR TRACKER</i>	
<i>COLD GAS PROPULSION SYSTEM</i>	
<i>HIGH RESOLUTION SUN SENSOR & ELECTRONICS</i>	
<i>(2) SUN SENSORS (HEADS) & ELECTRONICS</i>	
<i>INSTRUMENT DATA PROCESSING UNIT (DPU)</i>	

<i>STEREO 1 & 2 INSTRUMENTS</i>		<i>TIME</i>
■	<i>HELIOSPHERIC IMAGER (HI)</i>	<i>2 WKS.</i>
■	<i>ENERGETIC PARTICLE DETECTOR (EPD)</i>	<i>2 WKS.</i>
■	<i>RADIO BURST TRACKER (RBT)</i>	<i>2 WKS.</i>
■	<i>MAGNETOMETER & ELECTRONICS</i>	<i>2 WKS.</i>
■	<i>SOLAR WIND PLASMA ANALYZER (SWPA)</i>	<i>2 WKS.</i>
■	<i>SOLAR CORONAL IMAGING PACKAGE (SCIP)</i>	<i>2 WKS.</i>

STEREO 1 & 2 ORDNANCE AND THERMAL SYSTEMS

	<i>TIME</i>
■ <i>MECHANICAL AND ELECTRICAL TESTING OF ALL PYRO-ORDNANCE SYTEMS . (BOOMS, DEPLOYERS, ETC.)</i>	<i>1 WK.</i>
■ <i>INTEGRATE AND TEST THERMAL SYSTEM.. (INSTALL MULTI-LAYER INSULATION,ETC.)</i>	<i>1 WK.</i>
■ <i>STEREO 1 & 2 BASELINE ELECTRICAL PERFORMANCE AND DSN COMPATILBILITY TESTING (JPL DSN TRAILER)</i>	<i>2WKS.</i>
<i>TOTAL ESTIMATED TIME TO I & T (2) SPACECRAFT...30 WKS.</i>	
<i>(7 MONTHS)</i>	

ESTIMATED TIME TO ENVIRONMENTALLY TEST STEREO 1 & 2 SPACECRAFT

- | | | |
|---|--|-------------------------------|
| ■ | <i>PREPARATION FOR (2) SPACECRAFT VIBRATED IN SERIES. (MOVE TO VIBRATION FACILITY, INSTRUMENT WITH ACCELEROMETERS, FILL PROPULSION SYSTEM, MOUNT TO SHAKER, AND PERFORM PRE-VIBRATION ALIVENESS TESTS.</i> | <i>TIME
1 WK.
APL</i> |
| ■ | <i>VIBRATION TESTING (3 AXIS, WITH FUNCTIONAL TEST BETWEEN EACH AXIS, (2) SPACECRAFT IN SERIES).
PERFORM ALIGNMENT CHECKS PRE & POST VIBRATION.</i> | <i>2 WKS.
APL</i> |

*ENVIRONMENTALLY TESTING (2) STEREO
SPACECRAFT IN SERIES.*

ACOUSTIC AND SHOCK TESTING.

TIME

1WK.

- *SHIP (2) STEREO SPACECRAFT TO GSFC (GSFC)
UNPACK AND SETUP WITH GSE IN ACOUSTIC
CELL.*

- *PERFORM ACOUSTIC & SHOCK WITH PRE &POST 1 WK.
FUNCTIONAL ELECTRICAL TESTS AND PRE &POST (GSFC)
ALIGNMENT CHECKS.*

ENVIRONMENTALLY TESTING (2) STEREO SPACECRAFT IN SERIES.

- | | |
|--|----------------------------------|
| ■ <i>SETUP FOR MASS PROPS. AND SPIN BALANCE
TESTS IN ACOUSTIC CELL. (SPIN TABLE IN ACOUSTIC
CELL)</i> | <i>TIME
1 WK.
(GSFC)</i> |
| ■ <i>PERFORM MASS PROPS. AND SPIN BALANCE
TESTING WITH POST FUNCTIONAL ELECTRICAL TESTS.
AND ALIGNMENT CHECKS.</i> | <i>1 WK.
(GSFC)</i> |

*ENVIRONMENTALLY TESTING (2) STEREO
SPACECRAFT.*

- | | <i>TIME</i> |
|---|---------------|
| ■ <i>PREPARE AND SETUP (2) SPACECRAFT IN
THE THERMAL VACUUM CHAMBER AT GSFC.</i> | <i>1 WK.</i> |
| ■ <i>THERMOCOUPLES, HEATER SHROUDS,CABLING,
ETC.</i> | <i>(GSFC)</i> |
| ■ <i>PERFORM THERMAL VACUUM BALANCE AND CYCLING</i> | <i>4 WKS.</i> |
| ■ <i>TEST WITH FULL PERFORMANCE ELECTRICAL TESTS
PERFORMED ON EACH SPACECRAFT AS SIDE 1 (S/C 1)
AND SIDE 2 (S/C 2). GSE & GSS SUPPORTING EACH S/C</i> | <i>(GSFC)</i> |
| ■ <i>BREAK FROM CHAMBER, ALIGNMENTS, PACK & SHIP.</i> | <i>1 WK.</i> |
| ■ <i>TOTAL ESTIMATED ENVIRON. TEST TIME.... 13 WKS. (3 MONTHS)</i> | |

STEREO (2) SPACECRAFT INTEGRATED AND TESTED
AS ONE (WITH A SIDE 1 AND SIDE 2)

I &T STEREO S/C 1 (SIDE 1) & STEREO S/C 2 (SIDE 2)	ENVIRONMENTAL TESTING STEREO S/C 1 STEREO S/C 2
0 MONTHS	7/ 8 MONTHS 10